2 LSST System Design

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This chapter covers the basic elements of the LSST system design, with particular emphasis on those elements that may affect the scientific analyses discussed in subsequent chapters. We start with a description of the planned observing strategy in § 2.1, and then go on to describe the key technical aspects of system, including the choice of site (§ 2.2), the telescope and optical design (§ 2.3), and the camera including the characteristics of its sensors and filters (§ 2.4). The key elements of the data management system are described in § 2.5, followed by overviews of the procedures that will be invoked to achieve the desired photometric (§ 2.6) and astrometric (§ 2.7) calibration.

2.1 The LSST Observing Strategy

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The fundamental basis of the LSST concept is to scan the sky deep, wide, and fast with a single observing strategy, giving rise to a data set that simultaneously satisfies the majority of the science goals. This concept, the so-called "universal cadence," will yield the main deep-wide-fast survey (typical single visit depth of $r \sim 24.5$) and use about 90% of the observing time. The remaining 10% of the observing time will be used to obtain improved coverage of parameter space such as very deep ($r \sim 26$) observations, observations with very short revisit times (~ 1 minute), and observations of "special" regions such as the ecliptic, Galactic plane, and the Large and Small Magellanic Clouds. We are also considering a third type of survey, micro-surveys, that would use about 1% of the time, or about 25 nights over ten years.

The observing strategy for the main survey will be optimized for the homogeneity of depth and number of visits over 20,000 deg^2 of sky, where a "visit" is defined as a pair of 15-second exposures, performed back-to-back in a given filter, and separated by a four-second interval for readout and opening and closing of the shutter. In times of good seeing and at low airmass, preference is

given to r-band and i-band observations, as these are the bands in which the most seeing-sensitive measurements are planned. As often as possible, each field will be observed twice, with visits separated by 15-60 minutes. This strategy will provide motion vectors to link detections of moving objects, and fine-time sampling for measuring short-period variability. The ranking criteria also ensure that the visits to each field are widely distributed in position angle on the sky and rotation angle of the camera in order to minimize systematics that could affect some sensitive analyses, such as studies of cosmic shear.

The universal cadence will also provide the primary data set for the detection of near-Earth Objects (NEO), given that it naturally incorporates the southern half of the ecliptic. NEO survey completeness for the smallest bodies (~ 140 m in diameter per the Congressional NEO mandate¹) is greatly enhanced, however, by the addition of a crescent on the sky within 10° of the northern ecliptic. Thus, the "northern Ecliptic proposal" extends the universal cadence to this region using the r and i filters only, along with more relaxed limits on airmass and seeing. Relaxed limits on airmass and seeing are also adopted for ~ 700 deg² around the South Celestial pole, allowing coverage of the Large and Small Magellanic Clouds.

Finally the universal cadence proposal excludes observations in a region of 1,000 deg² around the Galactic Center, where the high stellar density leads to a confusion limit at much brighter magnitudes than those attained in the rest of the survey. Within this region, the Galactic Center proposal provides 30 observations in each of the six filters, distributed roughly logarithmically in time (it may not be necessary to use the bluest u and g filters for this heavily extincted region). The resulting sky coverage for the LSST baseline cadence, based on detailed operations simulations described in § 3.1, is shown for the r band in Figure 2.1. The anticipated total number of visits for a ten-year LSST survey is about 2.8 million (~ 5.6 million 15-second long exposures). The per-band allocation of these visits is shown in Table 1.1.

Although the uniform treatment of the sky provided by the universal cadence proposal can satisfy the majority of LSST scientific goals, roughly 10% of the time may be allocated to other strategies that significantly enhance the scientific return. These surveys aim to extend the parameter space accessible to the main survey by going deeper or by employing different time/filter sampling.

In particular, we plan to observe a set of "deep drilling fields," whereby one hour of observing time per night is devoted to the observation of a single field to substantially greater depth in individual visits. Accounting for read-out time and filter changes, about 50 consecutive 15-second exposures could be obtained in each of four filters in an hour. This would allow us to measure light curves of objects on hour-long timescales, and detect faint supernovae and asteroids that cannot be studied with deep stacks of data taken with a more spread-out cadence. The number, location, and cadence of these deep drilling fields are the subject of active discussion amongst the LSST Science Collaborations; see for example the plan suggested by the Galaxies Science Collaboration at § 9.8. There are strong motivations, e.g., to study extremely faint galaxies, to go roughly two magnitudes deeper in the final stacked images of these fields than over the rest of the survey.

These LSST deep fields will have widespread scientific value, both as extensions on the main survey and as a constraint on systematics. Having deeper data to treat as a model will reveal critical

¹H.R. 1022: The George E. Brown, Jr. Near-Earth Object Survey Act;

http://www.govtrack.us/congress/bill.xpd?bill=h109-1022



Figure 2.1: The distribution of the r band visits on the sky for one simulated realization of the baseline main survey. The sky is shown in Aitoff projection in equatorial coordinates and the number of visits for a 10-year survey is color-coded according to the inset. The two regions with smaller number of visits than the main survey ("minisurveys") are the Galactic plane (arc on the left) and the so-called "northern Ecliptic region" (upper right). The region around the South Celestial Pole will also receive substantial coverage (not shown here).

systematic uncertainties in the wider LSST survey, including photometric redshifts, that impact the measurements of weak lensing, clustering, galaxy morphologies, and galaxy luminosity functions.

A vigorous and systematic research effort is underway to explore the enormously large parameter space of possible survey cadences, using the Operations Simulator tool described in § 3.1. The commissioning period will be used to test the usefulness of various observing modes and to explore alternative strategies. Proposals from the community and the Science Collaborations for specialized cadences (such as mini-surveys and micro-surveys) will also be considered.

2.2 Observatory Site

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The LSST will be constructed on El Peñón Peak (Figure 2.2) of Cerro Pachón in the Northern Chilean Andes. This choice was the result of a formal site selection process following an extensive study comparing seeing conditions, cloud cover and other weather patterns, and infrastructure issues at a variety of potential candidate sites around the world. Cerro Pachón is located ten kilometers away from Cerro Tololo Inter-American Observatory (CTIO) for which over ten years of detailed weather data have been accumulated. These data show that more than 80% of the nights are usable, with excellent atmospheric conditions. Differential image motion monitoring (DIMM) measurements made on Cerro Tololo show that the expected mean delivered image quality is 0.67'' in g (Figure 2.3).



Figure 2.2: Artist's rendering of the LSST and dome enclosure on the summit of Cerro Pachón. The Auxiliary calibration telescope (§ 2.6) is also illustrated on a neighboring peak. (Image Credit: Michael Mullen Design, LSST Corporation.)

Cerro Pachón is also the home of the 8.2-m diameter Gemini-South and 4.3-m diameter Southern Astrophysical Research (SOAR) telescopes. Observations with those telescopes have confirmed the excellent image quality that can be obtained from this site. In addition, LSST will benefit from the extensive infrastructure that has been created on Cerro Pachón and La Serena to support these other facilities. The property is owned by the Association of Universities for Research in Astronomy (AURA), which also supports operation of CTIO, Gemini-South, and SOAR.

The LSST Observatory as a whole will be distributed over four sites: the Summit Facility on El Peñón, the Base Facility, the Archive Center, and the Data Centers. The Base Facility will be at the AURA compound in the town of La Serena, 57 km away from the mountain. The Archive Center will be at the National Center for Supercomputing Applications (NCSA) on the campus of the University of Illinois at Urbana-Champaign. There will be two Data Centers, one co-located with the Archive Center at NCSA, and one at the Base Facility in La Serena. Although the four facilities are distributed geographically, they are functionally connected via dedicated high-bandwidth fiber optic links.

2.3 Optics and Telescope Design

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Figure 2.3: The distribution of "seeing" (FWHM of the image of a point source) at 500 nm based on ten years of measurements from CTIO (10 km from the LSST site). The red curve shows results from a Differential Image Motion Monitor (DIMM), while the blue curve shows the delivered image quality. The mean is 0.67'', and the median is 0.59''.

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The LSST optical design shown in Figure 2.4 is a modified Paul-Baker three-mirror system (M1, M2, M3) with three refractive lenses (L1, L2, L3) and a color filter before the sensor at the focal plane. Conceptually, it is a generalization of the well-known Mersenne-Schmidt family of designs and produces a large field of view with excellent image quality (Wilstrop 1984; Angel et al. 2000; Seppala 2002). Spot diagrams are shown in the figure inset; these are made quantitative in Figure 2.5, which shows the encircled energy diameters at 50% and 80% in each filter as delivered by the baseline optical design. The uniformity across the field is striking.

The LSST étendue (including the effects of camera vignetting) is $319 \text{ m}^2 \text{deg}^2$. The effective focal length of the optical system is 10.3 m, making the final f/number 1.23. The plate scale is 50 microns per arcsecond at the focal surface. This choice of effective focal length represents an optimum balance of image sampling, overall system throughput, and manufacturing feasibility. The on-axis collecting area is 35 m^2 , equivalent to a 6.7-m diameter unobscured clear aperture.

The primary mirror (M1) is 8.4 m in diameter with a 5.1-m inner clear aperture. The tertiary mirror (M3) is 5 m in diameter. The relative positions of M1 and M3 were adjusted during the design process so that their surfaces meet with no axial discontinuity at a cusp, allowing M1 and M3 to be fabricated from a single substrate (see Figure 2.6). The 3.4-m convex secondary mirror (M2) has a 1.8-m inner opening. The LSST camera is inserted through this opening in order to access the focal surface.

The three reflecting mirrors are followed by a three-element refractive system that corrects field flatness and chromatic aberrations introduced by the filter and vacuum window. The 3.5° field of



Figure 2.4: The optical design configuration showing the telescope (left) and camera (right) layouts. Diffraction images in r for three field radii, 0, 1.0, and 1.75 degrees, are shown in boxes 0.6 arcseconds square (3 × 3 pixels).

view (FOV) covers a 64-cm diameter flat focal surface. Spectral filters reside between the second and third refractive lens as shown on the right side of Figure 2.4.

The image brightness is constant to a field radius of 1.2 degrees and gradually decreases afterward by about 10% at the 1.75-degree field edge. The intrinsic image quality from this design is excellent. The design also has very low geometrical distortion, with the distortion in scale $\Delta l/l < 0.1\%$ over the full FOV, making the LSST an excellent system for positional astrometry.

There are five aspheric surfaces in the optical design: each of the three mirror surfaces and one surface each on two of the camera lenses. The asphericity on the two concave surfaces of M1 and M3 are well within standard fabrication methods used for astronomical mirrors. During the optimization process, the asphericity of M2 was minimized to 18.9 microns of departure from the best-fit sphere in order to reduce the technical challenge for this optic. The three fused-silica refractive elements, which have clear apertures of 1.55 m, 1.10 m, and 0.72 m, while large, do not present any particular challenge in their fabrication. The 0.75-m diameter spectral filter is located just prior to L3. The filter thickness varies from 13.5 to 26.2 mm depending on the choice of spectral band, and is used to maintain the balance of lateral chromatic aberration. The zero-power meniscus shape of the filters keeps the filter surface perpendicular to the chief ray over the full field of view. This feature minimizes shifting of the spectral band wavelength with field angle. The last refractive element, L3, is used as the vacuum barrier to the detector cryostat. The central thickness of L3 is 60 mm to ensure a comfortable safety margin in supporting the vacuum stresses.

The proposed LSST telescope is a compact, stiff structure with a powerful set of drives, making it one of the most accurate and agile large telescopes ever built. The mount is an altitude over azimuth configuration (Figure 2.7). The telescope structure is a welded and bolted steel system designed to be a stiff metering structure for the optics and a stable platform for observing (Neill 2006, 2008). The primary and tertiary mirrors are supported in a single cell below the elevation ring; the camera and secondary mirror are supported above it. The design accommodates some ontelescope servicing as well as efficient removal of the mirrors and camera, as complete assemblies, for periodic maintenance.



Figure 2.5: The 50% (plain symbols) and 80% (symbols with lines) encircled energy diameter as a function of radius in the field of view for the LSST baseline optical design. The image scale is 50 microns per arcsec, or 180 mm per degree.



Figure 2.6: Design and dimensions of the primary and tertiary mirror, showing that the two are built out of a single mirror blank.

The stiffness of this innovative design is key to achieving a slew and settle time that is beyond the capability of today's large telescopes. The size and weight of the systems are a particular challenge, but the fast optical system allows the mount to be short and compact. Finite element analysis has been used to simulate the vibrational modes of the telescope system, including the concrete pier. The frequencies of the four modes with largest amplitudes are (in order):

- 8.3 Hz: Transverse telescope displacement;
- 8.7 Hz: Elevation axis rotation;
- 11.9 Hz: Top end assembly optical axis pumping; and
- 12.6 Hz: Camera pivot.

As described in § 2.1, the standard visit time in a given field is only 34 seconds, quite short for most telescopes. The time required to reorient the telescope must also be short to keep the fraction of time spent in motion below 20% (§ 1.6.2). The motion time for a nominal 3.5° elevation move and



Figure 2.7: Rendering of the telescope, showing mirror support structures, top end camera assembly, and integrated baffles.

a 7° azimuth move is five seconds. In two seconds, a shaped control profile will move the telescope, which will then settle down to less than 0.1'' pointing error in three seconds. The stiffness of the support structure and drive system has been designed to limit the amplitude and damp out vibrations at these frequencies within this time. The mount uses 400 horsepower in the azimuth drive system and 50 horsepower in the elevation system. There are four motors per axis configured in two sets of opposing pairs to eliminate hysteresis in the system. Direct drive systems were judged overly complicated and too excessive, so the LSST design has each motor working through a multi-stage gear reduction, with power applied through helical gear sets. The 300-ton azimuth assembly and 151-ton elevation assembly are supported on hydrostatic bearings. Each axis uses tape encoders with 0.001'' resolution. Encoder ripple from these tapes often dominates control system noise, so LSST will include adaptive filtering of the signal in the control loop. All-sky pointing performance will be better than 2''. Pointing will directly impact trailing and imaging systematics for LSST's wide field of view, so accurate pointing is key to tracking performance. Traditional closed loop guiding will achieve the final level of tracking performance.

2.4 Camera

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The LSST camera, shown in Figure 2.8, contains a 3.2-gigapixel focal plane array (Figure 2.9) comprised of 189 4 K × 4 K CCD sensors with 10 μ m pixels. The focal plane is 0.64 m in diameter, and covers 9.6 deg² field-of-view with a plate scale of 0.2" pixel⁻¹. The CCD sensors are deep depletion, back-illuminated devices with a highly segmented architecture, 16 channels each, that enable the entire array to be read out in two seconds (Figure 2.10).



Figure 2.8: Cutaway drawing of the LSST camera. The camera body is approximately 1.6 m in diameter and 3.5 m in length. The optic, L1, is 1.57 m in diameter.



Figure 2.9: With its 189 sensors, each a $4 \text{ K} \times 4 \text{ K}$ charge-coupled device (CCD), the focal plane of the camera images 9.6 deg² of the sky per exposure. Note the presence of wavefront sensors, which are fed back to the mirror support/focus system, and the guide sensors, to keep the telescope accurately tracking on a given field.



Figure 2.10: A schematic of the LSST sensor, showing the segmentation into 16 channels, each of which is read out in parallel.

The detectors are grouped into 3×3 arrays called "rafts." All the rafts are identical; each contains its own dedicated front-end and back-end electronics boards, which fit within the footprint of its sensors, thus serving as a 144-Megapixel camera on its own. The rafts and associated electronics are mounted on a silicon carbide grid inside a vacuum cryostat, with an intricate thermal control system that maintains the CCDs at an operating temperature of -100° C. The grid also contains two guide sensors and a wavefront sensor positioned at each of the four corners at the edge of the field. The entrance window to the cryostat is the third of three refractive lenses, L3 in Figure 2.8. The other two lenses, L1 and L2, are mounted in an "optics housing" at the front of the camera body. The camera body also contains a mechanical shutter and a filter exchange system holding five large optical filters, any of which can be inserted into the camera field of view for a given exposure. The system will in fact have six filters; the sixth filter can replace any of the five via an automated procedure accomplished during daylight hours.

2.4.1 Filters

The LSST filter complement (u, g, r, i, z, y) is modeled on the system used for the SDSS (Fukugita et al. 1996), which covers the available wavelength range with roughly logarithmic spacing while avoiding the strongest telluric emission features and sampling the Balmer break. Extension of the SDSS system to longer wavelengths (y-band) is possible because the deep-depletion CCDs have high sensitivity to 1 μ m (Figure 2.11).

The current LSST baseline design has a goal of 1% relative photometric calibration (§ 1.5), which

 Table 2.1:
 Design of Filters:
 Transmission Points in nanometers

Filter	Blue Side	Red Side	Comments
u	320	400	Blue side cut-off depends on AR coating
g	400	552	Balmer break at 400 nm
r	552	691	Matches SDSS
i	691	818	Red side short of sky emission at 826 nm
z	818	922	Red side stop before H_2O bands
y	950	1080	Red cut-off before detector cut-off



Figure 2.11: The left panel shows the transmission efficiency of the *ugrizy* filters by themselves as calculated from models of the filter performance. The total throughput, accounting for the transmission through the atmosphere at the zenith, the reflectivity of the reflective optics, the transmissivity of the refractive optics, and the quantum efficiency of the sensors is displayed in the panel on the right.

drives the requirements on the filter set. The filter set wavelength design parameters and the approximate FWHM transmission points for each filter are given in Table 2.1 and in Figure 2.11.

The filters consist of multi-layer dielectric interference coatings deposited on fused silica substrates. The baseline design has the first surface of the filters concentric about the chief ray in order to keep the angles of the light rays passing through the filters as uniform as possible over the entire range of field positions. The central thickness and the curvature of the second surface are optimized for image quality.

2.4.2 Sensors

The heart of the camera is the science sensor. Its key characteristics are as follows:

High quantum efficiency from 320 to 1080 nm. This is achieved using a large depletion depth $(100 \ \mu m)$ and implementation of the sensor in a back-illuminated configuration with a thin entrance window.

Minimal detector contribution to the point spread function. To reduce charge diffusion, the sensor is fully depleted, and a high internal field is maintained within the depletion region. This is made

possible by the use of high resistivity substrates, high applied voltages, and back-side contacts. Light spreading prior to photo-conversion at longer wavelengths is a minor contributor at the 100 μ m depletion depth.

Tight flatness tolerances. The fast LSST beam (f/1.23) yields a short depth of field, requiring $< 10 \ \mu m$ peak-to-valley focal plane flatness with piston, tip, and tilt adjustable to $\sim 1 \mu m$. This is achieved through precision alignment and mounting both within the rafts, and within the focal plane grid.

High fill factor. A total of 189 4 K × 4 K sensors are required to cover the 3200 cm² focal plane. To maintain high throughput, the sensors are mounted in four-side buttable packages and are positioned in close proximity to one another with gaps of less than a few hundred μ m. The resulting "fill factor," i.e., the fraction of the focal plane covered by pixels, is 93%.

Fast readout. The camera is read out in two seconds. To reduce the read noise associated with higher readout speeds, the sensors are highly segmented. The large number of I/O connections then requires that the detector electronics be implemented within the cryostat to maintain a manageable number of vacuum penetrations.

Our reference sensor design is a CCD with a high degree of segmentation, as illustrated in Figure 2.10. A $4 \text{ K} \times 4 \text{ K}$ format was chosen because it is the largest footprint consistent with good yield. Each amplifier will read out 1,000,000 pixels (a 2000×500 sub-array), allowing a pixel readout rate of 500 kHz per amplifier. The sensors are mounted on aluminum nitride (AlN) packages. Traces are plated directly to the AlN insulator to route signals from the CCD to the connectors on the back of the package. The AlN package provides a stiff, stable structure that supports the sensor, keeps it flat, and extracts heat via a cooling strap.

2.4.3 Wavefront Sensing and Guiding

Four special purpose rafts, mounted at the corners of the science array, contain wavefront sensors and guide sensors (Figure 2.9). Wavefront measurements are accomplished using curvature sensing, in which the spatial intensity distribution of stars is measured at equal distances on either side of focus. Each curvature sensor is composed of two CCD detectors, with one positioned slightly above the focal plane, the other positioned slightly below the focal plane. The CCD technology for the curvature sensors is identical to that used for the science detectors in the focal plane, except that the curvature sensor detectors are half-size so they can be mounted as an in-out defocus pair. Detailed analyses have verified that this configuration can reconstruct the wavefront to the required accuracy. These four corner rafts also hold two guide sensors each. The guide sensors monitor the locations of bright stars at a frequency of ~ 10 Hz to provide feedback for a loop that controls and maintains the tracking of the telescope at an accurate level during an exposure. The baseline sensor for the guider is the Hybrid Visible Silicon hybrid-CMOS detector. We have carried out extensive evaluation to validate that its characteristics (including wide spectral response, high fill factor, low noise, and wide dynamic range) are consistent with guiding requirements.

2.5 Data Management System

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The LSST Data Management System ("DMS") is required to generate a set of data products and to make them available to scientists and the public. To carry out this mission the DMS performs the following major functions:

- Continually processes the incoming stream of images generated by the camera system during observing to produce transient alerts and to archive the raw images.
- Roughly once per year², creates and archives a Data Release ("DR"), which is a static selfconsistent collection of data products generated from all survey data taken from the date of survey initiation to the cutoff date for the Data Release. The data products include optimal measurements of the properties (shapes, positions, fluxes, motions) of all objects, including those below the single visit sensitivity limit, astrometric and photometric calibration of the full survey object catalog, and limited classification of objects based on both their static properties and time-dependent behavior. Deep coadded images of the full survey area are produced as well.
- Periodically creates new calibration data products, such as bias frames and flat fields, that will be used by the other processing functions.
- Makes all LSST data available publicly through an interface and databases that utilize, to the maximum possible extent, community-based standards such as those being developed by the Virtual Observatory ("VO"), and facilitates user data analysis and the production of user-defined data products at Data Access Centers and at external sites.

The geographical layout of the DMS facilities is shown in Figure 2.12; the facilities include the Mountain Summit/Base Facility at Cerro Pachón and La Serena, the central Archive Center at NCSA, the Data Access Centers at NCSA and La Serena, and a System Operations Center. The data management system begins at the data acquisition interface between the camera and telescope subsystems and flows through to the data products accessed by end users. On the way, it moves through three types of managed facilities supporting data management, as well as end user sites that may conduct science using LSST data or pipeline resources on their own computing infrastructure.

- The data will be transported over existing high-speed optical fiber links from the Mountain Summit/Base Facility in Chile to the archive center in the U.S. Data will also flow from the Mountain Summit/Base Facility and the archive center to the data access centers over existing fiber optic links. The Mountain Summit/Base Facility is composed of the mountaintop telescope site, where data acquisition must interface to the other LSST subsystems, and the Base Facility, where rapid-turnaround processing will occur for data quality assessment and near real-time alerts.
- The Archive Center is a super-computing-class data center with high reliability and availability. This is where the data will undergo complete processing and re-processing and permanent storage. It is also the main repository feeding the distribution of LSST data to the community.

²In the first year of operations, we anticipate putting out data releases every few months.



Figure 2.12: A schematic map of the LSST DMS facilities. The LSST Telescope Site, located on Cerro Pachón, Chile, is connected to the Base Facility, located in La Serena, Chile by a dedicated fiber link. The Base Facility is connected to the Archive Center, located at the National Center for Supercomputing Applications in Illinois using commercial high-speed network links. The Archive Center, in turn, fans out data to Data Access Centers which serve the data to clients, and may be located anywhere in the world. The System Operations Center monitors and controls the overall operation of the DMS facilities, and provides end-user support facilities.

- Data Access Centers for broad user access are envisioned, according to a tiered access model, where the tiers define the capacity and response available especially to computationally expensive queries and analyses. There are two project-funded Data Access Centers co-located with the Base Facility and the Archive Center. These centers provide replication of all of the LSST data to ensure that disaster recovery is possible. They provide Virtual Observatory interfaces to the LSST data products. LSST is encouraging non-US/non-Chilean funding for potential partner institutions around the world to host additional Data Access Centers, which could increase end user access bandwidth, provide local high-end computation, and help amortize observatory operations costs.
- The System Operations Center provides a control room and large-screen display for supervisory monitoring and control of the DM System. Network and facility status are available as well as the capability to "drill down" to individual facilities. The Center will also provide DM support to observatory science operations, as well as an end user help desk.

2.5.1 LSST Data Product Overview

Level 1, 2, and 3 Data Products

The data products are organized into three groups, based largely on where and when they are produced.

- Level 1 products are generated by pipeline processing the stream of data from the camera system during normal observing. Level 1 data products are, therefore, continuously generated and/or updated every observing night. This process is of necessity highly automated, and must proceed with absolutely minimal human interaction. Level 1 products include *alerts*, i.e., announcements that the flux or position of a given object has changed significantly relative to the long-term average. The alerts will be released within 60 seconds of the closing of the shutter at the end of a visit (§ 1.5). In addition to science data products, a number of Level 1 science data quality assessment ("SDQA") data products are generated to assess quality and to provide feedback to the Observatory Control System.
- Level 2 products are generated as part of a yearly Data Release. Level 2 products use Level 1 products as input, and include data products for which extensive computation is required (such as variability information, detection, and measurement of the properties of faint objects, and so on), often because they combine information from the stack of many exposures. Although the steps that generate Level 2 products will be automated, significant human interaction may be required at key points to ensure the quality of the data.
- Level 3 data products are derived from Level 1 and/or Level 2 data products to support particular science goals, often requiring the combination of LSST data across significant areas on the sky. The DMS will facilitate the creation of Level 3 data products, for example by providing suitable Applications Programming Interfaces (APIs) and computing infrastructure, but is not itself required to create any Level 3 data product. Instead these data products are created externally to the DMS, using software written by, for example, science collaborations. Once created, Level 3 data products may be associated with Level 1 and Level 2 data products through database federation³. In rare cases, the LSST Project, with the agreement of the Level 3 creators, may decide to incorporate Level 3 data products into the DMS production flow, thereby promoting them to Level 2 data products.

Level 1 and Level 2 data products that have passed quality control tests will be accessible to the public without restriction. Additionally, the source code used to generate them will be made available, and LSST will provide support for building the software system on selected platforms. The access policies for Level 3 data products will be product- and source-specific, and in some cases will be proprietary.

Overview of Pipeline Processing

In the overall organization of the DMS pipelines and productions, "production" has a particular meaning: it is a coordinated group of pipelines that together carry out a large-scale DMS function.

Alert Production Astronomers interested in transient phenomena of many sorts (Chapter 8) need to know of objects whose flux has changed significantly as soon as possible after the data are taken. Therefore, the most visible aspect of Level 1 processing is the production of *alerts*, i.e., announcements of such variability. The Alert Production is directly fed by the output data stream from the camera Science Acquisition System (SDS) during observing. This data stream contains

³See Wikipedia's article on the subject at http://en.wikipedia.org/wiki/Federated_database.

both unprocessed (raw) camera images, and images that have been corrected for crosstalk by the SDS on the mountain. At the end of a visit, the Alert Production:

- Acquires the raw science images from the camera, and moves them to the Archive Center for permanent storage.
- Processes the crosstalk-corrected images from the camera to detect transient events within 60 seconds of shutter closure for the second exposure in a visit. This will probably be done with a variant of the Alard & Lupton (1998) image-subtraction algorithm.
- Packages catalog information, together with postage-stamp images of detected transients as alerts and past history of the object, and distributes them to the community as VO events.
- Continuously assesses the data quality of the data stream.

The major steps in the processing flow are:

- Image processing of the raw exposures to remove the instrumental signature, such as bias, flat-field, bad columns, and so on.
- Determination of the World Coordinate System (WCS), image Point-Spread Function (PSF), and rough photometric zeropoint. This produces processed exposures.
- Subtraction of a registered template exposure (a co-addition of previous high-quality images of a given field, created in an earlier data release) from the processed exposure, producing a difference exposure.
- Detection of sources (both positive and negative!) in the difference exposure, producing what we refer to hereafter as "DIASources."
- Visit processing logic, which compares the DIASources from the two exposures in the visit to discriminate against cosmic rays, and to flag very rapidly moving Solar System objects.
- "FaintSources," abbreviated measurements of low signal-to-noise ratio (S/N) detections, are produced for objects of particular interest, e.g., predicted positions for Solar System objects, or objects that have previously produced *alerts*.
- Comparison of positive flux DIASources with predictions from the Moving Object Pipeline (MOPS; see § 2.5.3) for already known Solar System objects, as contained in the Moving Object table.
- The Association Pipeline is run to match DIASources to already known astronomical objects, as contained in the Object table.
- DIASources that are detected in both exposures of a visit, and are not matched to a known Solar System object, produce an *alert*.
- Quality Assessment is performed at every pipeline stage, stored in database tables, and fed to the Observatory Control System as required.
- The Moving Object Pipeline (§ 2.5.3) is run during the day to interpret each new detection of a moving object as a new measurement of a Solar System object already in the Moving Object table, or as a previously unknown object, which will be added to the Moving Object table. All orbits are refined based on the new measurements from the night.

The community has strongly expressed the preference that *alerts* not be significantly filtered prior to distribution so that science opportunities are not closed off. We have, therefore, adopted very simple criteria for issuing an *alert*: 5σ DIASources seen in both exposures of a visit which are not consistent with cosmic ray events.

Note that no explicit classification of an *alert* is provided, but users can readily construct classifiers and filters based on information in the Science Database; indeed, this is likely to be part of Level 3 software produced by the transient, stellar populations, and supernova science collaborations. The information that could be used for this classification includes the light curve, colors, and shape information for the associated object. Additionally, database queries can readily be formulated which will identify exposures that have generated anomalously large numbers of *alerts*, presumably due to image artifacts or processing problems.

As the raw images arrive at the Archive Center, the same processing flow is performed there, with the consistency of the databases at the Base and Archive Centers being periodically checked. The duplication of processing is carried out to reduce the data bandwidth required between the Base and Archive Centers.

Data Release Production At yearly intervals (more often during the first year of the survey) a new Data Release (DR) is produced. A DR includes all data taken by the survey from day one to the cutoff date for the DR, and is a self-contained set of data products, all produced with the same pipeline software and processing parameters. The major steps in the processing flow are:

- As in the Alert Production, all raw exposures from the camera are processed to remove the instrumental signature, and to determine the WCS and PSF, producing processed exposures. This is done with the best available calibration products, which in general will be superior to those available when the processing was initially done.
- The survey region is tessellated into a set of sky patches of order the size of a CCD, and several co-added exposures are produced for each patch from the processed exposures. These are a per-band template co-add used for image subtraction; a detection co-add used in the Deep Detection Pipeline (see next item), possibly per-band; and a RGB co-add used for visualization.
- The Deep Detection Pipeline is run, populating the Object, Source, and FaintSource tables. Rather than working from the co-add, Deep Detection will use the "Multifit" algorithm (§ 2.5.2; Tyson et al. 2008), whereby a model (e.g., a PSF for a stellar object or an exponential profile for a disk galaxy) is fit to the entire stack of exposures which contain the object. Thus each exposure is fit using its own PSF; this results in a set of optimal measurements of the object attributes over the full time span of the survey, including astrometric parameters such as proper motion and parallax.
- The Image Subtraction Pipeline is run, as in the Alert Production, yielding DIASources and FaintSources for transient objects.
- The Moving Object Pipeline is run on DIASources, to yield a complete set of orbits for Solar System Objects in the Moving Object table.

• The Photometric Calibration Pipeline is run on the full set of measurements in the Source, DIASource, and FaintSource catalogs, incorporating measurements from the Auxiliary Telescope and other sources of data about the atmosphere to perform a global photometric calibration of the survey (§ 2.6). In addition to accurate photometry for every measurement, this yields an atmosphere model for every exposure.

2.5.2 Detection and Measurement of Objects

Here we provide more detail on the specific algorithms used to define and measure object properties that are issued with the Data Releases:

Deep Detection Processing

The survey region is organized into overlapping sky patches of order the size of a CCD, and a deep co-added image is created for each patch. The details of the co-add algorithm are still undecided, but the current baseline is to use the Kaiser (2004) algorithm on the full stack of survey images contained within the Data Release. The Kaiser algorithm convolves each image with the reflection of its PSF, and then accumulates with weight inversely proportional to the sky variance. Care will be taken to ensure that rapidly moving objects, such as Solar System objects, do not appear in the co-add. An object detection algorithm is then run on the co-add, generating an initial Object catalog. An "Object" at this stage is nothing more than a pixel footprint on the sky, possibly with links to related Objects in a segmentation tree that has been created by segmenting (deblending) overlapping Objects. The tree will be organized so that the root node is the largest object in the hierarchy, with the leaf nodes being the smallest. The segmentation/deblending algorithm to be employed is still under investigation, with Sextractor (Bertin & Arnouts 1996) or the SDSS photometric pipeline (Stoughton et al. 2002) being examples of the kind of processing involved. The properties of the Objects that are segmented in this way are then determined with Multifit as described below.

Difference Exposure Processing

A new object is created whenever a transient source that is detected in both difference images from a visit does not match any object already in the table. The match will take account of extendedness as well as position on the sky, so that a new point source at the location of a galaxy already in the catalog (for example, due to a supernova or variable AGN) will result in a new object.

Note that this process cannot be perfect, since measuring the extendedness of objects near the PSF size will always be uncertain. Consequently, there will be cases where flux from a supernova or AGN point source will be incorrectly added to the underlying galaxy rather than to a new point source. Between successive Data Releases, however, these errors will decrease in severity: As the survey goes deeper, and accumulates images in better seeing, extendedness will be better measured by the Multifit procedure, as discussed below.

Measuring the Properties of Objects

The image pixels containing an object from all relevant exposures are fit to one or more object models using Multifit, generating model parameters and a covariance matrix. Our choice of models is driven by astrophysics, by characteristics of the LSST system, and by computing practicalities. The initial model types are as follows:

Slowly Moving Point Source Model. The Slowly Moving Point Source (SMPS) Model is intended to account for the time varying fluxes and motion on the sky of point sources (usually stars) with proper motions between zero and roughly $10'' \,\mathrm{yr}^{-1}$. The model accounts for motion with respect to the local astrometric reference frame that is generated by proper motion, parallax, and possibly orbital motion with respect to a binary companion. The object properties are measured in every exposure that contains it. If the S/N in the exposure is above a predetermined threshold, perhaps 5, the measurement generates a row in the Source table. If the S/N is lower than the threshold, a FaintSource row is generated instead. Lang et al. (2009) have successfully used a similar modeling and measurement approach to detect very faint brown dwarfs with high proper motion.

The SMPS Model will be fit only to objects that are leaf nodes in the segmentation tree.

Small Object Model. The Small Object (SO) Model is intended to provide a robust parametrization of small (diameter < 1') galaxy images for weak lensing shear measurement and determination of photometric redshifts. The definition of the model flux profile is still undecided (Sersic profiles? Superpositions of exponential and de Vaucouleurs profiles?), but should be driven by the needs of photometric redshifts (§ 3.8). The measurement of the elliptical shape parameters will be driven by the needs of weak lensing (Chapter 14).

The SO Model will be fit only to objects that are leaf nodes in the segmentation tree.

Large Object Model. A "large" object is one for which the 20 mag/arcsec² isophotal diameter is greater than 1′, and less than 80% of the patch size. This includes, for example, the majority of NGC galaxies. The vast majority of the LSST science will be accomplished with measurements made using the SMPS and SO Models, but much valuable science and numerous EPO applications will be based on larger objects found in LSST images. To at least partially satisfy this need, large objects will have entries in the Object table, but will not have any model fitting performed by Multifit.

Solar System Model. The predicted ephemerides from the orbit for an object in the moving object table constitutes an object model which is used to measure the object properties in each exposure that contains the object. It is not yet decided whether the measurements of faint detections should be at a position entirely fixed by the orbit prediction, or should be allowed to compensate for prediction error by "peaking up" within some error bound around the prediction.

The Multifit Algorithm

Objects are detected on co-added images, but their models will be fit to the full data set of exposures on which they appear ($n \sim 400$ at the end of the survey in each filter). The motivation for doing this is two-fold (Tyson et al. 2008). First, the co-add will have a very complicated and

discontinuous PSF and depth patchiness due to detector gaps and masked moving objects. Second, although the Kaiser co-add algorithm is a sufficient statistic for the true sky under the assumptions that sky noise dominates, and is Gaussian, those assumptions do not strictly hold in real data.

An initial model will be fit to the co-add, to provide a good starting point for the fit to the full data set. Multifit will then read in all the pixels from the n exposures and perform a maximum likelihood fit for the model which, when convolved with the n PSFs, best matches the n observations. This naturally incorporates the effects of varying seeing, as the contribution of the better-seeing images to the likelihood will be sharper. This approach also facilitates proper accounting for masked areas, cosmic rays, and so on. The best-fit model parameters and their uncertainties will be recorded in an Object table row.

Model Residuals

The measurement process will produce, in conjunction with every source, a residual image that is the difference of the associated image pixels and the pixels predicted from the model over the footprint of the model. Characterizing these residuals is important for science such as strong lensing and merging galaxies, that will identify interesting candidates for detailed analysis through their residuals. Selecting the most useful statistical measures of the residuals will be the outcome of effort during the continuing design and development phase of the project.

2.5.3 The Moving Object Processing System (MOPS)

Identifying moving objects and linking individual detections into orbits, at all distances and solar elongations, would be a daunting task for LSST without advanced software. Each observation from the telescope is differenced against a "template" image (built from many previous observations), allowing detection of only transient, variable, or moving objects in the result. These detections are fed into the Moving Object Processing System (MOPS), which attempts to link these individual detections into orbits.

MOPS uses a three-stage process to find new moving objects (Kubica 2005; Kubica et al. 2005, 2007). In the first stage, intra-night associations are proposed by searching for detections forming linear "tracklets." By using loose bounds on the linear fit and the maximum rate of motion, many erroneous initial associations can be ruled out. In the current model of operations, LSST will revisit observed fields twice each night, with approximately 20–45 minutes between these observations. These two detections are what are linked into tracklets. In the second stage, internight associations are proposed by searching for sets of tracklets forming a quadratic trajectory. Again, the algorithm can efficiently filter out many incorrect associations while retaining most of the true associations. However, the use of a quadratic approximation means that a significant number of spurious associations still remains. Current LSST operations simulations (§ 3.1) show that LSST will image the entire visible night sky approximately every three nights - thus these inter-night associations of "tracklets" into "tracks" are likely to be separated by 3–4 nights.

In the third stage, initial orbit determination and differential corrections algorithms (Milani et al. 2008) are used to further filter out erroneous associations by rejecting associations that do not correspond for a valid orbit. Each stage of this strategy thus significantly reduces the number

of false candidate associations that the later and more expensive algorithms need to test. After orbit determination has occurred, each orbit is checked against new or previously detected (but unlinked) tracklets, to extend the orbit's observational arc.

To implement this strategy, the LSST team has developed, in a collaboration with the Pan-STARRS project (Kaiser et al. 2002), a pipeline based on multiple k-dimensional- (kd-) tree data structures (Kubica et al. 2007; Barnard et al. 2006). These data structures provide an efficient way to index and search large temporal data sets. Implementing a variable tree search we can link sources that move between a pair of observations, merge these tracklets into tracks spread out over tens of nights, accurately predict where a source will be in subsequent observations, and provide a set of candidate asteroids ordered by the likelihood that they have valid asteroid tracks. Tested on simulated data, this pipeline recovers 99% of correct tracks for near-Earth and main belt asteroids, and requires less than a day of CPU time to analyze a night's worth of data. This represents a several thousand fold increase in speed over a naïve linear search. It is noteworthy that comparable amounts of CPU time are spent on the kd-tree based linking step (which is very hard to parallelize) and on posterior orbital calculations to weed out false linkages (which can be trivially parallelized).

2.5.4 Long-term Archive of LSST Data

The LSST will archive all observatory-generated data products during its entire 10-year survey. A single copy of the resultant data set will be in excess of 85 petabytes. Additional scientific analyses of these data have the potential to generate data sets that significantly exceed this amount.

The longer-term curation plan for the LSST data beyond the survey period is not determined, but it is recognized as a serious concern. This issue is important for all large science archives and it is impractical (perhaps impossible) for individual facilities or researchers to address this problem unilaterally.

The NSF has recognized this issue and has begun soliciting input for addressing long-term curation of scientific data sets via the DataNet and other initiatives. The LSST strongly endorses the need for this issue to be addressed at the national level, hopefully via a partnership involving government, academic, and industry leaders.

2.6 Photometric Calibration

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2.6.1 Natural LSST Photometric System

A ground-based telescope with a broad-band detector will observe the integral of the source specific flux density at the top of the Earth's atmosphere, $F_{\nu}(\lambda)$, weighted by the normalized response function (which includes the effects of the atmosphere and all optical elements), $\phi_b(\lambda)$,

$$F_b = \int_0^\infty F_\nu(\lambda)\phi_b(\lambda)d\lambda,$$
(2.1)

where the index b corresponds to a filter bandpass (b = ugrizy). The chosen units for F_b are Jansky (1 Jansky = 10^{-26} W Hz⁻¹ m⁻² = 10^{-23} erg cm⁻² s⁻¹ Hz⁻¹), and by definition, $\int_0^\infty \phi_b(\lambda) d\lambda = 1$. The corresponding astronomical magnitude is defined as

$$m_b \equiv -2.5 \log_{10} \left(\frac{F_b}{F_{AB}}\right). \tag{2.2}$$

The flux normalization $F_{AB} = 3631$ Jy follows the standard of Oke & Gunn (1983).

The normalized response function is defined as

$$\phi_b(\lambda) \equiv \frac{\lambda^{-1} T_b(\lambda)}{\int_0^\infty \lambda^{-1} T_b(\lambda) d\lambda}.$$
(2.3)

The λ^{-1} term reflects the fact that the CCDs used as sensors in the camera are photon-counting devices rather than calorimeters. Here, $T_b(\lambda)$ is the system response function,

$$T_b(\lambda) = T_b^{instr}(\lambda) \times T^{atm}(\lambda), \qquad (2.4)$$

where T^{atm} is the optical transmittance from the top of the atmosphere to the input pupil of the telescope, and T_b^{instr} is the instrumental system response ("throughput") from the input pupil to detector (including filter b). This function is proportional to the probability that a photon starting at the top of the atmosphere will be recorded by the detector. Note that the overall normalization of both T_b^{instr} and T^{atm} cancels out in Equation 2.3.

An unavoidable feature of ground-based broad-band photometry is that the normalized response function, $\phi_b(\lambda)$, varies with time and position on the sky and detector due to variations in shapes (spectral profiles) of $T^{atm}(\lambda)$ and $T_b^{instr}(\lambda)$. Traditionally, these effects are calibrated out using a set of standard stars. Existing data (e.g., from SDSS) demonstrate that this method is insufficient to deliver the required photometric precision and accuracy in general observing conditions. Instead, the LSST system will measure $T^{atm}(\lambda)$ and $T_b^{instr}(\lambda)$ (yielding measured quantities S^{atm} and S_b^{instr}) on the relevant wavelength, temporal, and angular scales.

In summary, the basic photometric products will be reported on a *natural photometric system*, which means that for each photometric measurement, F_b^{meas} , a corresponding measured normalized response function, $\phi_b^{meas}(\lambda)$, will also be available. Of course, error estimates for both F_b^{meas} and $\phi_b^{meas}(\lambda)$ will also be reported. The survey will collect ~ 10¹² such ($F_b^{meas}, \phi_b^{meas}$) pairs over a ten year period – one pair for each source detection.

2.6.2 Standardized Photometric System

One of the fundamental limitations of broad-band photometry is that measurements of flux, F_b^{meas} , cannot be accurately related to $F_{\nu}(\lambda)$ unless $\phi_b(\lambda)$ is known. An additional limitation is that F_b^{meas} can vary even when $F_{\nu}(\lambda)$ is constant because ϕ_b is generally a variable quantity. This variation

needs to be accounted for, for example, when searching for low-amplitude stellar variability, or construction of precise color-color and color-magnitude diagrams of stars.

Traditionally, this flux variation is calibrated out using atmospheric extinction and color terms, which works for sources with relatively smooth spectral energy distributions. However, strictly speaking this effect cannot be calibrated out unless the shape of the source spectral energy distribution,

$$f_{\nu}(\lambda) = F_{\nu}(\lambda)/F_0, \qquad (2.5)$$

where F_0 is an arbitrary normalization constant, is known. If $f_{\nu}(\lambda)$ is known, then for a predefined "standard" normalized response function, $\phi_b^{std}(\lambda)$ (obtained by appropriate averaging of an ensemble of ϕ_b^{meas} during the commissioning period), the measurements expressed on the natural photometric system can be "standardized" as

$$m_b^{std} - m_b^{meas} \equiv \Delta m^{std} = 2.5 \log \left(\frac{\int_0^\infty f_\nu(\lambda)\phi_b^{meas}(\lambda)d\lambda}{\int_0^\infty f_\nu(\lambda)\phi_b^{std}(\lambda)d\lambda} \right),\tag{2.6}$$

where we have used magnitudes for convenience. While this transformation is in principle exact, m_b^{std} inherits measurement error in m_b^{meas} , as well as an additional error due to the difference between the true $\phi_b(\lambda)$ and the measured ϕ_b^{meas} which will be used in practice. Uncertainties in our knowledge of $f_{\nu}(\lambda)$ will contribute an additional error term to m_b^{std} . Depending on the science case, users will have a choice of correcting m_b^{meas} using pre-computed Δm^{std} for typical spectral energy distributions (various types of galaxies, stars, and solar system objects, average quasar spectral energy distribution, etc.), or computing their own Δm^{std} for their particular choice of $f_{\nu}(\lambda)$.

2.6.3 Measurement of Instrumental System Response, S_b^{sys}

A monochromatic dome projector system will be used to provide a well-controlled source of light for measurement of the relative throughput of the full LSST instrumental system. This includes the reflectivity of the mirrors, transmission of the refractive optics and filters, the quantum efficiency of the sensors in the camera, and the gain and linearity of the sensor read-out electronics.

An array of projectors mounted in the dome of the LSST enclosure will be illuminated with both broadband (e.g., quartz lamp) and tunable monochromatic light sources. These "flat-field" projectors are designed to fill the LSST étendue with uniform illumination, and also to limit stray light emitted outside the design acceptance of the system. A set of precision diodes will be used to normalize the photon flux integrated during flat-field exposures. These photodiodes, together with their read-out electronics, will be calibrated at the U.S. National Institute of Standards (NIST) to $\sim 0.1\%$ relative accuracy across wavelength from 450 nm to 950 nm. The response of these diodes varies smoothly across this range of wavelength and provides a well-behaved reference (Stubbs 2005). Adjustment of the wavelength of the light source can be as fine as one nanometer, and will allow precise monitoring of the shape of the bandpasses of the instrumental system during the course of the survey (Stubbs & Tonry 2006).

It is anticipated that the shapes of the bandpasses will vary only slowly, so detailed measurement will need be done only once per month or so. But build-up of dust on the surfaces of the optics will occur more rapidly. The dimensions of these particles are generally large, and their shadows will be out of focus at the focal plane. So the loss of throughput due to them will be independent of wavelength - i.e., "gray", and the pixel-to-pixel gradients of their shadows will not be large. Daily broadband and "spot-checks" at selected wavelengths with the monochromatic source will be used to measure day-to-day changes in the system passbands.

2.6.4 Measurement of Atmospheric Transmittance, S^{atm}

Many studies have shown that atmospheric transmission can be factored into the product of a frequency dependent ("non-gray") part that varies only on spatial scales larger than the telescope field-of-view and temporal scales long compared with the interval between LSST exposures; and a frequency independent part ("gray" cloud cover) that varies on moderately short spatial scales (larger than the PSF) and temporal scales that may be shorter than the interval between exposures:

$$S^{atm}(alt, az, t, \lambda) = S^{atm}_{q}(alt, az, t) \times S^{atm}_{nq}(alt, az, t, \lambda).$$
(2.7)

The measurement strategies to determine S_g^{atm} and S_{ng}^{atm} are quite different:

- S_{ng}^{atm} is determined from repeated spectroscopic measurements of a small set of probe stars by a dedicated auxiliary telescope.
- S_g^{atm} is determined from the LSST science images themselves, first approximately as each image is processed, and later more precisely as part of a global photometric self-calibration of the survey. The precise measurement of S_g^{atm} is based on the measured fluxes of a very large set of reference stars that cover the survey area and are observed over many epochs. Every exposure contains a large enough set of sufficiently stable stars that a spatial map can be made of S_q^{atm} across each image.

The LSST design includes a 1.2-m auxiliary calibration telescope located on Cerro Pachón near the LSST that will be used to measure $S_{ng}^{atm}(alt, az, t, \lambda)$. The strategy is to measure the full spatial and temporal variation in atmospheric extinction throughout each night independently of operations of the main survey telescope. This will be done by repeatedly taking spectra of a small set of probe stars as they traverse the sky each night. These stars are spaced across the sky to fully cover the area surveyed by the LSST main telescope. The calibration will use state-of-the-art atmospheric models (Stubbs et al. 2007) and readily available codes (MODTRAN4) to accurately compute the signatures of all significant atmospheric components in these spectra. This will allow the atmospheric mix present along any line of sight at any time to be interpolated from the measured data. The probe stars will be observed many times during the LSST survey, so the SED of each star can be bootstrapped from the data. The instrumental response of the spectrograph can also be bootstrapped from the data by including stars with a variety of SEDs over a broad range of airmass.

2.6.5 Calibration Procedure

Two levels of LSST calibration will be carried out at differing cadences and with differing performance targets. A nightly data calibration based on the best available set of prior calibrated observations will provide "best-effort" precision and accuracy. This calibration will be used for quality assurance, generation of alerts to transients, and other quantities appropriate for Level 1 Data Products (§ 2.5.1). A more complete analysis will recalibrate the data accumulated by the survey at periodic "Data Release" dates (Level 2 in the terminology of § 2.5.1). It is this repeated calibration of the accumulated survey that will be held to the survey requirements for photometric repeatability, uniformity, and accuracy.

LSST photometric calibration is then separated into three parts that address different science requirement specifications:

- Relative calibration: normalization of internal (instrumental) measurements in a given bandpass relative to all other measurements made in the same bandpass across the sky.
- Absolute calibration of colors: determination of the five unique differences between flux normalizations of the six bands (color zero points).
- Absolute calibration of flux: definition of the overall physical scale of the LSST magnitude system, i.e., normalization to F_{AB} in Equation 2.2.

Relative Calibration

Precision relative calibration of LSST photometry will be accomplished by analysis of the repeated observations of order 10^8 selected bright (17 < r < 20) isolated stars during science operations of the survey. The LSST image processing pipelines will extract raw ADU counts for these stars from each image, and the data release Calibration Pipeline will process data from the calibration auxiliary subsystems to determine the optical bandpass appropriate for each image. These measurements will be used to determine calibrations for all sources detected on each image.

After reduction of each image in the accumulated survey, the Calibration Pipeline will execute a global self-calibration procedure that will seek to minimize the dispersion of the errors in all observations of all reference stars. This process is based on techniques used in previous imaging surveys (Glazebrook et al. 1994; MacDonald et al. 2004), and the specific implementation used by LSST will be based on the "Übercal" procedure developed for SDSS (Padmanabhan et al. 2008). "Calibration patches" of order the size of a single CCD will be defined on the camera focal plane. The LSST survey will dither pointings from epoch to epoch to control systematic errors, so stars will fall on different patches on different epochs across the sky. The measured magnitudes of reference stars will be transformed (Equation 2.6) to the LSST standard bandpass using the accumulated estimates of the colors of each star and the corresponding measured observational bandpasses. The Calibration Pipeline will minimize the relative error $\delta_b(p, j)$ in the photometric zero-point for each patch, p, on each image, j, of the accumulated survey by minimizing,

$$\chi^2 = \sum_{(i,j)} \frac{\left(m_b^{std,meas}(i,j) - \left(m_b^{std,true}(i) + \delta_b(p,j)\right)\right)^2}{\left(\delta m_b^{std,meas}(i,j)\right)^2},\tag{2.8}$$

where the magnitudes are in the standard system, and the summation is over all stars, i, in all images, j. These $\delta_b(p, j)$ will be used to correct the photometry for all other sources in patch, p, on image, j.

Absolute Calibration of Colors and Flux

There are six numbers, Δ_b , for the entire survey that set the zeropoints of the standard bandpasses for the six filters. These six numbers can be expressed in terms of a single fiducial band, which we take to be the r band,

$$\Delta_b = \Delta_r + \Delta_{br}.\tag{2.9}$$

The LSST strategy to measure the observational bandpass for each source is designed to reduce errors, Δ_{br} , in the five color zero points, to meet specifications in the survey requirements. This process will be validated with the measured flux from one or more celestial sources, most likely hot white dwarfs whose simple atmospheres are reasonably well-understood.

At least one external flux standard will be required to determine Δ_r (one number for the whole survey!). While one celestial standard would be formally sufficient, choosing a number of such standards would provide a powerful test for Δ_r . Identification of such a standard, or set of standards, has not yet been done.

2.7 Astrometric Calibration

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The astrometric calibration of LSST data is critical for many aspects of LSST operations (pointing, assessment of camera stability, etc.) and scientific results ranging from the measurement of stellar parallaxes and proper motions to proper performance of difference image analysis.

The core of the astrometric algorithm is the simultaneous solution for two types of unknowns, the coefficients that transform the coordinates on the focal plane measured in a given exposure into some common coordinate system (absolute astrometry), and the positions and motions of each star (relative astrometry). Whereas a direct solution exists, it involves the inversion of relatively large matrices and is rarely used. Instead, the solution is based on an iterative improvement given the prior knowledge of positions of a relatively small number of stars (from a reference catalog or similar). All observations for all stars in a small area of sky are extracted from the database. Using the catalog positions for the stars as a first guess, the transformations from each observation to the catalog system are computed, and then all measures for each star are used to compute the new values for position and motion.

2.7.1 Absolute Astrometry

The current realization of the International Celestial References Frame (ICRF) is defined by the stars in ESA's Hipparcos mission catalog. ESA's Gaia mission, set to launch in 2012, will improve the ICRS and ICRF by another two orders of magnitude down to the level of a few micro-arcseconds.

Absolute calibration consists of computing the positions of all the detected sources and objects in the LSST imaging with respect to the ICRF. Were no improved catalogs available between now and LSST commissioning, the reference catalogs would be the US Naval Observatory's UCAC-3 catalog for bright optical stars (down to about 16th magnitude, uncomfortably close to LSST's saturation limit) or the NASA 2MASS catalog whose near-IR positions for optical stars have an accuracy of 70-100 milli-arcseconds (mas) for individual stars and systematic errors in the range of 10-20 mas. There are large numbers of 2MASS stars in each and every LSST field of view, so the astrometric calibration is little more than the computation of a polynomial that maps position on the focal plane into the system of right ascension and declination defined by the measured positions of catalog objects. The transformation is encapsulated in the World Coordinate System (WCS) keywords in the Flexible Image Transport System (FITS) header for each image.

One of the key astrometric challenges in generating and using these WCS solutions is the distinction between "observed" and "catalog" coordinates. When LSST takes an image, the stars and galaxies are at their observed positions. These positions include the astrometric effects of proper motion, parallax, differential refraction, differential aberration, and others. Most applications work in catalog coordinates such as the J2000 positions for objects or the equivalent for image manipulation. The astrometric calibration will provide a rigorous method for going between these coordinate systems.

2.7.2 Differential Astrometry

Differential astrometry is for most science the more important job to be done. The differential solution, which provides measures for the stellar parallax, proper motion, and perturbations (e.g., due to binary companions), can be substantially more accurate than the knowledge of the absolute coordinates of an object. The task is to measure centroids on images and to compute the transformation from the current frame into the mean coordinate system of other LSST data, such as the deep image stacks or the different images from the multi-epoch data set. The photon noise limit in determining the position of the centroiding of a star is roughly half the FWHM of the seeing disk, divided by the signal-to-noise ratio of the detection of the star. The expectation is that atmospheric seeing will be the dominant source of astrometric error for sources not dominated by photon statistics. Experiments with wide-field imaging on the Subaru Telescope (§ 3.6) suggest that accuracy will be better than 10 mas per exposure in the baseline LSST cadence, although it may be worse with objects with unusual SEDs such that simple differential color refraction analysis fails, or for exposures taken at extreme zenith angles.

Perhaps the biggest unknown in discussion of differential astrometry is the size of the "patch" on the sky over which the astrometric solution is taken. If the patch is small enough, the astrometric impact of the unaveraged turbulence can be mapped with a simple polynomial, and the differential astrometric accuracy approaches that set by the photon statistics. Our current understanding of atmospheric turbulence suggests that we will be able to work with patches between a few and 10 arcmin in size, small enough that the geometry can fit with low-order spatial polynomials. The current approach is to use the JPL HEALPix tessellation strategy. For each solution HEALPix(el), separate spatial transformations are computed for each CCD of each observation. These produce measures for each object in a mean coordinate system, and these measures can be fit for position, proper motion, parallax, refraction, perturbations from unseen companions, and other astrometric signals. Given the very faint limiting magnitude of LSST, there should be be a sufficient number of astrometrically useful galaxies to deliver a reasonable zero-point within each HEALpix⁴. The

 $^{^{4}}$ Quasars will be less useful; they are less numerous, and their very different SEDs cause different refraction from stars.

characterization of the zero-point errors and the astrometric utility of galaxies will be the major work area for the astrometric calibration team.

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